

1 Supplementary Materials

2 S1. Vehicle Dynamics

3 The dynamics of the vehicle on an incline with slope γ , which is a localized representation
 4 of substrate under the vehicle helps explain the acceleration's dependence on the head-
 5 ing θ and local tilting angle γ (Fig. S1a) in experiments. On the incline, we denote the
 6 direction along the gravity as \parallel and the direction perpendicular to it as \perp so that the
 7 acceleration from the gravity field is $a_{\perp}^g = 0, \sim a_{\parallel}^g = g \sin \gamma$ (Fig. S1b). Considering this
 8 incline as a localized picture of the vehicle's immediate substrate, here $\hat{\perp}$ direction stands
 9 for the $\hat{\phi}$ and $\hat{\parallel}$ direction stands for the \hat{r} .

10

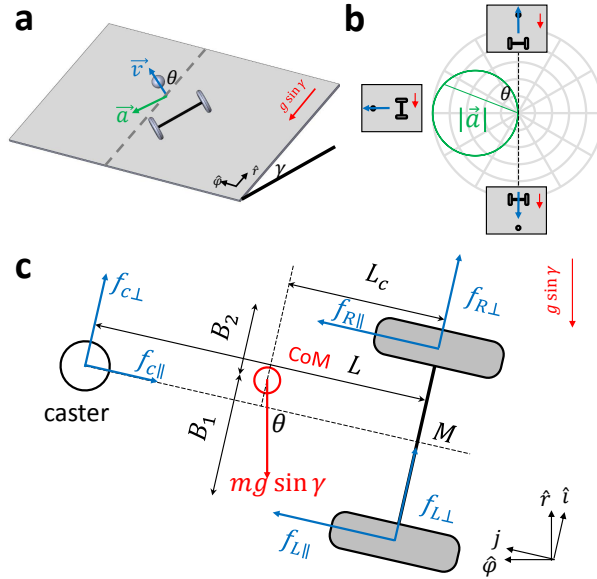


Figure S1: Vehicle dynamics of the robotic vehicle. (a) Modelling the dynamics of the vehicle on a slope with incline angle equalled to its current tilt γ . (b) The magnitude of the acceleration changes with the heading angle θ and vanishes when going along the gradient of the incline. (c) The force diagram of the vehicle.

11 Since the friction on the rolling caster is much smaller than the other friction forces, the
 12 vehicle rotates about the middle point of the wheel axis, M . The torque about M consists

13 of the frictions on the two wheels and the caster, as well as the gravity component in the
 14 plane. Since the two wheels are connected to a differential drive, the torques generated
 15 by the friction parallel to the wheel $f_{L\parallel}, f_{R\parallel}$ are of same magnitude and opposite signs
 16 and therefore cancelled out. The torques generated by the friction perpendicular to the
 17 wheel are zero since the forces pass through M .

18 The non-zero torques left with us are the one generated by the gravity component in
 19 the plane and the friction from the caster f_c :

$$\begin{aligned} \tau &= \left(\Delta B \hat{i} + L_c \hat{j} \right) \times \\ &\quad mg \sin \gamma (-\sin \theta \hat{i} - \cos \theta \hat{j}) + L \hat{j} \times f_{c\perp} \hat{i} \end{aligned} \quad (1)$$

$$= (mg \sin \gamma (-\Delta B \cos \theta + L_c \sin \theta) - f_{c\perp} L) \hat{k} \quad (2)$$

20 where $\Delta B \equiv \frac{1}{2}(B_2 - B_1)$.

21 The moment of inertia of the vehicle with respect to M is $I = I_{\text{vehicle}} + m(L^2 + \Delta B^2)$
 22 where we approximate $I_{\text{vehicle}} = \frac{1}{2}mR_v^2$ with R_v being the radius of the vehicle since
 23 the mass distribution is quite homogeneous. Therefore the magnitude of the angular
 24 acceleration ($\beta = \tau \cdot \hat{k}/I$) and the acceleration of the center of mass on the incline is

$$a_{\text{incline}} = \left| L_c \hat{j} + \Delta B \hat{i} \right| \cdot \beta \quad (3)$$

$$\approx L_c \cdot \frac{\tau \cdot \hat{k}}{I} \quad (4)$$

$$= \frac{mg \sin \gamma (L_c \sin \theta - \Delta B \cos \theta) - f_c L}{\frac{1}{2}mR_v^2 + mL_c^2 + m\Delta B^2} L_c \quad (5)$$

25 For the ideal case that the center of mass is not biased to the left or right so that
 26 $B_1 = B_2$, the acceleration is

$$\begin{aligned} a_{\text{incline}} &= \frac{mgL_c \sin \gamma \sin \theta - f_c L}{\frac{1}{2}mR_v^2 + mL_c^2} L_c \\ &= \frac{L_c^2}{\frac{1}{2}R_v^2 + L_c^2} g \sin \gamma \sin \theta - \frac{f_c L}{\frac{1}{2}mR_v^2 + mL_c^2} \end{aligned} \quad (6)$$

27 When $\theta = \pi/2$ and f_c being very small since this is a rolling friction, the acceleration
 28 projected onto the horizontal plane is

$$\begin{aligned} a(\theta = \pi/2) &= a_{\text{incline}}(\theta = \pi/2) \cos \gamma \\ &\approx \frac{L_c^2}{\frac{1}{2}R_v^2 + L_c^2} g \sin \gamma \cos \gamma \end{aligned} \quad (7)$$

29 The actual numbers in the experiment $R_v = 5$ cm, $L_c \approx 1$ cm give the theoretical
 30 prediction

$$a_{\text{theo}}(\theta = \pi/2) \approx 0.074 g \sin \gamma \cos \gamma \quad (8)$$

31 which is quite close to the experimental measurement

$$a_{\text{expt}}(\theta = \pi/2) = (0.073 \pm 0.001) g \sin \gamma \cos \gamma \quad (9)$$

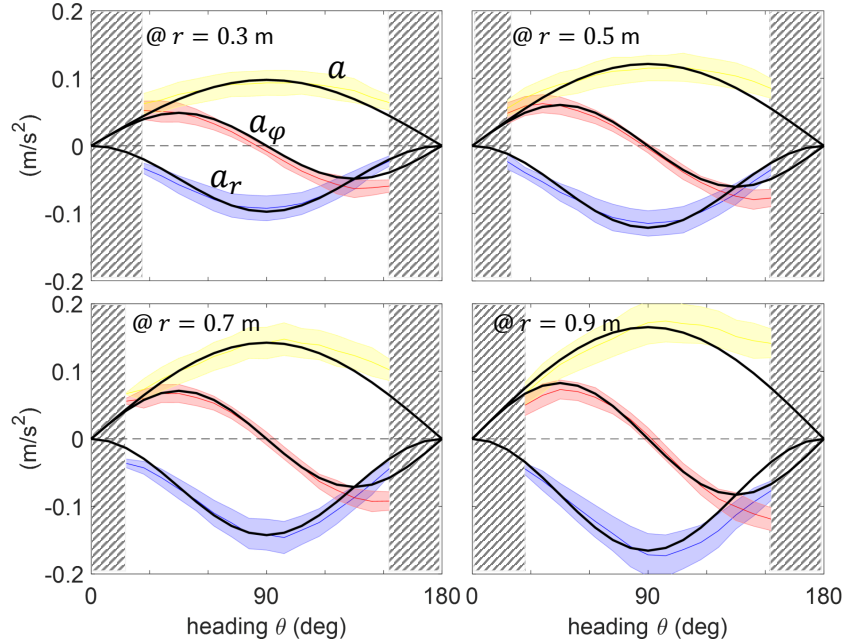


Figure S2: Acceleration at different radii. The shading colored in yellow, red and blue are the magnitude a , and the azimuthal, radial components a_φ , a_r of the acceleration respectively. The solid lines and shading in the figures denote the mean and standard deviation over 238 experiments. The black lines are the theory $a = k(r) \cdot \sin \theta$, $a_r = -a \sin \theta$, and $a_\varphi = a \cos \theta$ where $k(r)$ takes the mean value shown in Fig.3c in the main text.

32 In reality, there is always a small bias between B_1 and B_2 , this small correction from
 33 the CoM (center of mass) offset that breaks the symmetry of acceleration with respect to
 34 the heading gives the attraction to the circular orbit and will be discussed in Section S3.

35 This bias is

$$a_{\text{bias}} = -g \sin \gamma \cos \theta \frac{L_c \Delta B}{\frac{1}{2} R^2 + L_c^2 + \Delta B^2} \quad (10)$$

36 where ΔB can be measured by weighing the normal force on the left and right wheels
 37 and given by

$$\Delta B = \frac{L_w}{2} \frac{N_R - N_L}{N_R + N_L} \quad (11)$$

38 where N_L, N_R are the normal forces on the two wheels and $L_w = 6$ cm. For an imbalance
 39 of $(N_R - N_L)/(N_R + N_L) \approx 20$ % measured from experiment, it can be inferred that
 40 $\Delta B \approx 0.6$ cm. Thus, the maximum bias ($\theta = 0^\circ, 180^\circ$) when driving on a typical local
 41 slope of $\gamma = 10^\circ$ is $a_{\text{bias}} = 0.074$ m/s², which is about 40 % of the maximum magnitude of
 42 the acceleration in the system. Fig.S3 shows how this bias causes the slight dependence
 43 on θ .

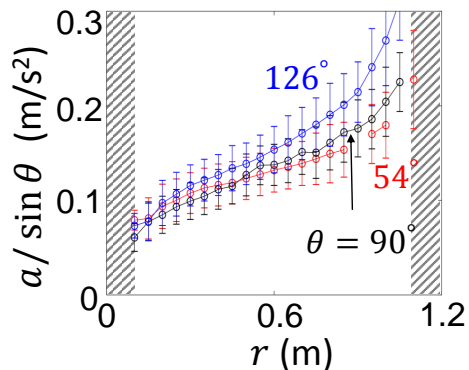


Figure S3: Plots of k as a function of r for various values of θ using $a/\sin \theta$. The gray shaded regions refer to regions which are forbidden due to steric exclusion.

44 S2. Transient Dynamics of a Vehicle with Slight Chirality

45 S2.1 Result

46 The transient behavior of some trajectories that decay into circular orbits is caused by
 47 the slight asymmetry in the mechanical structure that the center of mass (CoM) deviates
 48 slightly from the center-line. As shown in Section S1, the acceleration magnitude $|a|$ for
 49 a vehicle with slight asymmetry with respect to the heading is given by

$$|a| = k(r) \cdot (\sin \theta + \epsilon \cdot \cos \theta) \quad (12)$$

50 where $\epsilon = -\frac{\Delta B}{L_c}$ increases with the CoM's deviation from the center-line being ΔB .

51 This leads to the polar equation of the trajectory

$$\begin{aligned} r_{,\varphi\varphi} = \frac{2r^2}{r} + r - \tilde{k}(r) \cdot (r^2 + r_{,\varphi}^2) \\ - \epsilon \cdot \tilde{k}(r) \cdot (r_{,\varphi}r + \frac{r^3}{r}) \end{aligned} \quad (13)$$

52 where $\tilde{k} \equiv k/v^2$. Detailed derivation can be found in S2.2.

53 Let $r = r_c + \rho$ where ρ is the perturbation and r_c is the radius of the circular orbit
 54 that $k(r_c) = v^2/r_c$. After discarding the $O(\rho^2)$ terms, the differential equation is reduced
 55 to

$$\rho_{,\varphi\varphi} = -(1 + r_c k'_c/k_c)\rho - \epsilon\rho_{,\varphi} \quad (14)$$

56 where $k_c \equiv k(r_c)$, $k'_c \equiv k'(r_c)$.

57 The solution to this damped oscillator gives the solution as

$$\rho(\varphi) = \rho(0) \cos\left(\sqrt{1 + r_c k'_c/k_c - (\epsilon/2)^2}\varphi\right) e^{-\epsilon\varphi/2} \quad (15)$$

58 with an exponentially decaying envelope with a half-life $(2 \log 2)/\epsilon$ that degrades with the
 59 bias; that is, the larger the imperfection is, the faster the trajectory is attracted a circular
 60 orbit.

61 On the other hand, when the vehicle has an acceleration bias towards the orbit di-
62 rection, ϵ will be negative, then ρ will expand and leads the orbit to either crash to the
63 center or escape from the membrane. From this example with clockwise trajectory, we see
64 that the orbit is attracted to a circular orbit when $\epsilon \propto (B_2 - B_1) > 0$, that is when the
65 CoM is biased to the left wheel. The data listed in Section S1 shows an estimate $\epsilon \approx 0.5$,
66 indicating a half life of $(2 \log 2)/0.5 \approx 3$. This qualitatively matches with our experimen-
67 tal observation of the transient orbits when the vehicle tested on a leveled ground does
68 not drive sufficiently straight. We posit the quantitative difference may come from the
69 inaccuracy of the ΔB and L_c estimate.

70 In summary, a counterclockwise (clockwise) orbit will get attracted to a circular orbit
71 when the CoM is biased to the right (left) while the eccentricity increases to escape or
72 crash when the CoM is biased to the left (right).

73

74 2.2 Derivation

We consider a slightly simpler case where the membrane is rather flat that $\Psi^2 = 1 +$
 $(\partial z / \partial r)^2 \approx 1$. The acceleration components in radial and azimuthal directions are given
by Eqs.1, 2 in the main text as

$$\begin{cases} r\ddot{\varphi} + 2\dot{r}\dot{\varphi} = a_\varphi = f & (16a) \\ \ddot{r} - r\dot{\varphi}^2 = a_r = -f \cdot \tan \theta & (16b) \end{cases}$$

75 where $f = k(r) \cdot (\sin \theta + \epsilon \cos \theta) \cos \theta$ for a vehicle with bias ϵ .

76 The definition of the heading θ gives

$$\tan \theta \equiv \frac{v_\varphi}{v_r} = \frac{r\dot{\varphi}}{\dot{r}}. \quad (17)$$

77 Using $\dot{\varphi} = \dot{r} \tan \theta / r$, we get the time derivatives of azimuth φ as

$$\dot{\varphi} = \frac{\dot{r} \tan \theta}{r}, \quad \ddot{\varphi} = \frac{\ddot{r} \tan \theta}{r} + \frac{\dot{r} \dot{\theta} \sec^2 \theta}{r} - \frac{\dot{r}^2 \tan \theta}{r^2}. \quad (18)$$

78 Substitute the $\dot{\varphi}$ and $\ddot{\varphi}$ in (16) with (18) and eliminate \ddot{r} by (16a)-(16b) $\cdot \tan \theta$, we have

79

$$\dot{r}\dot{\theta} + \frac{\dot{r}^2}{r} \tan \theta = f. \quad (19)$$

Consider the radial speed as the velocity's projection on the radial direction $\dot{r} = v \cdot \cos \theta$, we arrive at the vector field description:

$$\begin{cases} \dot{r} = v \cdot \cos \theta & (20a) \\ \dot{\theta} = \frac{f(r, \theta)}{v \cdot \cos \theta} - \frac{v \cdot \sin \theta}{r}. & (20b) \end{cases}$$

Plug in $f = k(r) \cdot (\sin \theta + \epsilon \cos \theta) \cos \theta$, we have

$$\begin{cases} \dot{r} = v \cdot \cos \theta & (21a) \\ \dot{\theta} = (k/v - v/r) \sin \theta + (k/v)\epsilon \cos \theta. & (21b) \end{cases}$$

80 Divide (21a) by (21b), we have

$$\frac{dr}{d\theta} = \frac{v \cdot \cos \theta}{(k(r)/v - v/r + (k(r)/v)\epsilon \cos \theta) \cdot \sin \theta}. \quad (22)$$

81 As we want r to be a function of the azimuth φ , we convert all θ to φ . We use
 82 the definition of heading again $\tan \theta = r\dot{\varphi}/\dot{r} = (rd\varphi/dt)/(dr/dt) = rd\varphi/dr = r/r_{,\varphi}$,
 83 $\sin \theta = r/\sqrt{r^2 + r_{,\varphi}^2}$, and $\cos \theta = r_{,\varphi}/\sqrt{r^2 + r_{,\varphi}^2}$. The left hand side of (22) can thus be
 84 converted to

$$LHS = \frac{dr}{d\theta} = \frac{1}{\theta_r} = \frac{1}{\frac{d(\arctan(r/r_{,\varphi}))}{d(r/r_{,\varphi})} \cdot \frac{d(r/r_{,\varphi})}{dr}} = \frac{1}{[1 + (r/r_{,\varphi})^2]^{-1} \cdot \left[\frac{1}{r_{,\varphi}} - \frac{r \cdot r_{,\varphi\varphi}}{r_{,\varphi}^3} \right]} \quad (23)$$

85 The right hand side can be converted to

$$RHS = \frac{r_{,\varphi}}{(\tilde{k} - 1/r) \cdot r + \epsilon \tilde{k} r_{,\varphi}}. \quad (24)$$

86 where $\tilde{k}(r) \equiv k(r)/v^2$.

87 Equate the LHS (23) and the RHS (24), we finally arrive at

$$r_{,\varphi\varphi} = \frac{2r^2}{r} + r - \tilde{k}(r) \cdot (r^2 + r_{,\varphi}^2) - \epsilon \cdot \tilde{k}(r) \cdot (r_{,\varphi} r + \frac{r_{,\varphi}^3}{r}). \quad (25)$$

88

89 **S3. Trajectory resulting from active vehicle deviates from spatial-**
90 **only geodesics**

91 To measure the spatial-only trajectory, we let the left and right wheel speed of the vehicle
92 be the same. The spatial-only trajectory (blue) enabled by the same left and right wheel
93 speeds is much straighter than that of an active vehicle responding the local gradient
94 (red).

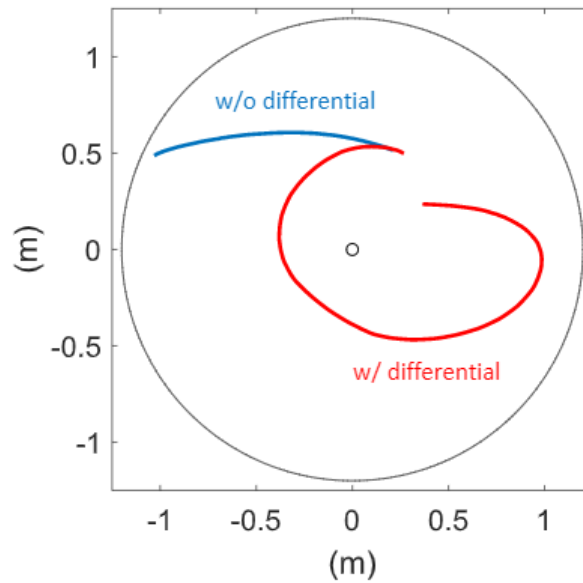


Figure S4: Comparison of the trajectories of the vehicle on the membrane when differential mechanism is applied and disabled. To disable the differential mechanism so that the two wheels are rigidly connected, the gears in the differential are glued.

95 S4. Conserved quantities

96 The metric $ds^2 = -\alpha^2 dt^2 + \Phi^2(\Psi^2 dr^2 + r^2 d\varphi^2)$ gives

$$-1 = -\alpha^2 \dot{t}^2 + \Phi^2(\Psi^2 \dot{r}^2 + r^2 \dot{\varphi}^2) \quad (26)$$

97 where $dq/d\lambda \equiv \dot{q}$ and we specify the affine parameter λ to be s . To convert the \dot{q} to \dot{q} ,
 98 we use $\dot{t} = E/\alpha^2$ and $\dot{\varphi} = L/\Phi^2 r^2$ from the conserved quantities in geodesic equations (see
 99 ‘Robophysical modeling of spacetime dynamics (arXiv:2202.04835)’ for details). Using
 100 $\dot{t} = E/\alpha^2$ again in \dot{r} , we have $\dot{r} = dr/d\lambda = (dr/dt)(dt/d\lambda) = \dot{r}\dot{t} = (E/\alpha^2)\dot{r}$. Plug these
 101 into Eq.26, we have

$$-1 = -\frac{E^2}{\alpha^2} + \Phi^2 \Psi^2 \frac{E^2 \dot{r}^2}{\alpha^4} + \frac{L^2}{\Phi^2 r^2} \quad (27)$$

102 Multiply both sides with $-\alpha^2/E^2$ and rearrange the terms, we arrive at

$$1 = \frac{\Phi^2}{\alpha^2} \Psi^2 \dot{r}^2 + \frac{1}{r^2} \frac{\alpha^2 L^2}{\Phi^2 E^2} + \frac{\alpha^2}{E^2} \quad (28)$$

103 which leads to Eq.4 in the main text.

104 To show the maximum of ℓ is obtained at r_0 , we plug the derived metric into Eq.??,??,

$$\ell \equiv \frac{L}{E} = e^{-K(r_0)/v^2} r_0 \cdot v \quad (29)$$

105 The maximum of ℓ is obtained at r_0 that

$$\frac{\partial \ell}{\partial r_0} = e^{-K(r_0)/v^2} \left(1 - \frac{r_0 k(r_0)}{v^2} \right) = 0, \quad (30)$$

106 showing r_0 coincides with the circular orbit radius r_c such that $k(r_c) = v^2/r_c$.

107 **S5. Generalization from axi-symmetric substrate to general sub-**
 108 **strate**

109 We construct the general model by viewing the terrain gradient in the axi-symmetric case
 110 as an arbitrary terrain gradient. The following table shows the comparison between the
 111 special case with axi-symmetry and the general case.

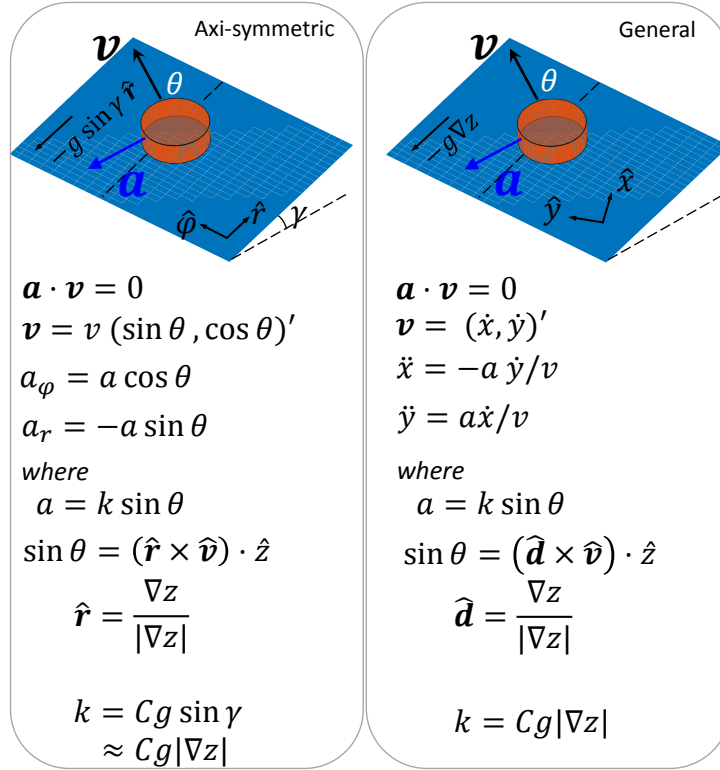


Figure S5: Generalization of the vehicle dynamics on an arbitrary terrain.

112 We construct the general dynamics by making analogy such that the axi-symmetric
 113 case is a special case of the general case. The analogies can be found in Fig.S5.

114 If we plug the generalized direction and magnitude into the acceleration components,
 115 we get

$$\ddot{x} = -a \frac{\dot{y}}{v} \quad (31)$$

$$= k \sin \theta \frac{\dot{y}}{v} \quad (32)$$

$$= k(\hat{\mathbf{d}} \times \hat{\mathbf{v}}) \cdot \hat{\mathbf{z}} \frac{\dot{y}}{v} \quad (33)$$

$$= Cg|\nabla z| \left(\frac{\nabla z}{|\nabla z|} \times \hat{\mathbf{v}} \right) \cdot \hat{\mathbf{z}} \frac{\dot{y}}{v} \quad (34)$$

$$= Cg(\nabla z \times \frac{\mathbf{v}}{v}) \cdot \hat{\mathbf{z}} \frac{\dot{y}}{v} \quad (35)$$

$$= \frac{Cg}{v^2}(z_{,x}\dot{y} - z_{,y}\dot{x})\dot{y} \quad (36)$$

$$= Cg\dot{y}(d_x\dot{y} - d_y\dot{x})/v^2 \quad (37)$$

116 Similarly, we have $\ddot{y} = -Cg\dot{x}(d_x\dot{y} - d_y\dot{x})/v^2$.

117 In both cases, the acceleration magnitude vanishes when the velocity is along the
 118 radial (gradient) direction and the acceleration direction is perpendicular to the velocity.

119 **S6. Membrane Measurement**

120 **S6.1 Membrane constant**

121 To model the membrane deformation, we consider a free circular membrane with radius R
 122 only deformed by its self weight and pressed by a cap in the center with depth D and cap
 123 radius $R_0 < R$. When the load from self weight is uniform, the height of the membrane
 124 z follows

$$\Delta Z = \lambda^{-1} \quad (38)$$

125 where λ absorbed the elasticity and the mass density.

126 Applying the axi-symmetry ($\partial Z/\partial\varphi = 0$) and boundary conditions $Z(R) = 0, Z(R_0) =$
 127 $-D$ for a membrane without a load such as the robotic vehicle, the general solution to a
 128 membrane deformed by only self weight is

$$Z(r) = \frac{1}{4\lambda}r^2 + C_1 \log r + C_2 \quad (39)$$

129 where

$$C_1 = \frac{D - \frac{1}{4\lambda}(R^2 - R_0^2)}{\log(R/R_0)}, \quad (40)$$

$$C_2 = \frac{\frac{1}{4\lambda}(R^2 \log R_0 - R_0^2 \log R) - D \log R}{\log(R/R_0)} \quad (41)$$

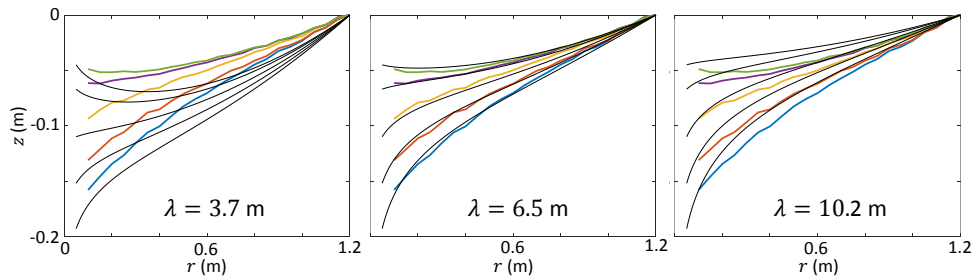


Figure S6: Membrane constant measurement: The black lines show the radial profiles of the free membrane from Poisson equation Eq.39. The colored lines show the measurement from experiments.

130 We measured the cross sections of the membrane with various central depressions D 's
 131 and compare them with solution Eq.39 for various λ . The value of λ is chosen such that
 132 the solutions match with experiments the best. In our setup, λ is measured to be 6.5 m
 133 (Fig.S6).

134 S6.2 Membrane isotropy

135 Ideally, the height of the membrane at a particular radius should be the same for any
 136 azimuthal angle in terms of the axi-symmetry. To understand how the membrane deviates
 137 from the ideal, the variation of this height is evaluated with the data taken from the
 138 Optitrack cameras for three different central depressions. The variation is found to be
 smaller than 5% of the central depression.

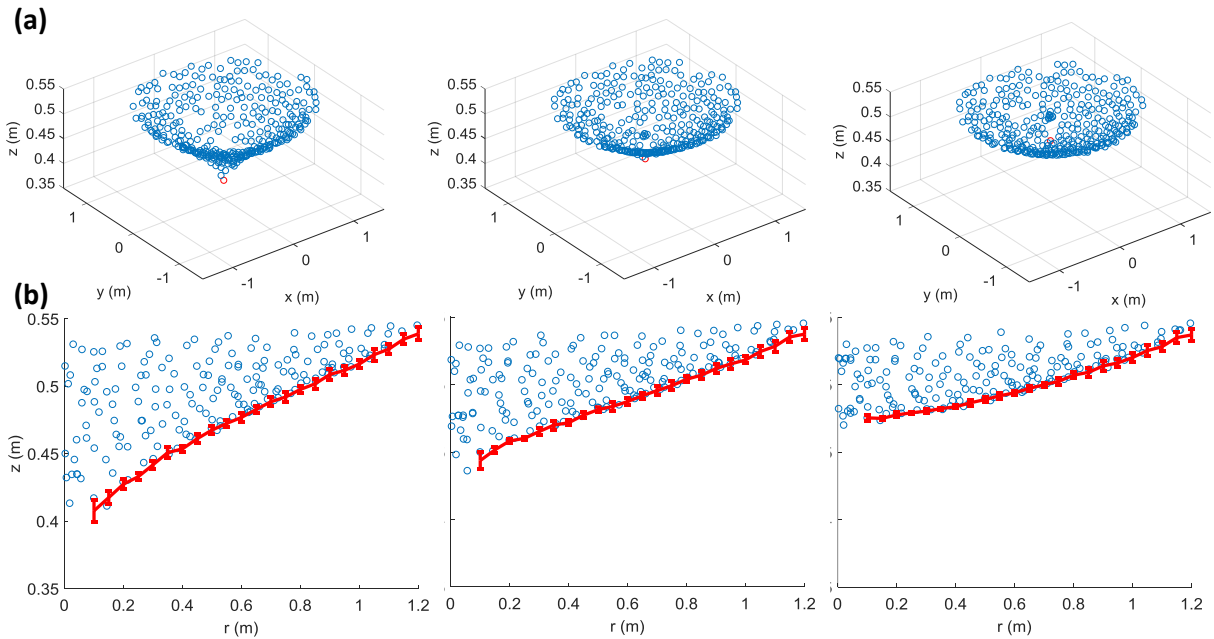


Figure S7: Shapes of the membrane with different central depressions. (a) The perspective views of the membrane profile measured from the optical tracking system. (b) The red curves show the heights averaged over the azimuthal angles.

139

140 **S7. Analytic solution to the membrane**

141 As shown in the previous section, the deformation of the membrane by its self weight can
 142 be well characterized by $\Delta Z = \lambda^{-1}$. To model the additional load from the vehicles besides
 143 the weight of the membrane itself, we evaluate the area density of vehicle and scaled it
 144 by that of the membrane so that $\Delta Z = \lambda^{-1}(1 + \tilde{P})$ with $\tilde{P} = \sigma_v/\sigma$ where σ_v and σ are
 145 the density of the vehicle ($\approx 20,000$ g/m²) and the membrane (137 g/m²) respectively.
 146 For simplicity, we assume the load is a uniform distribution on a disc centered at the i th
 147 vehicle's position \mathbf{r}_i and with the radius of the vehicle R_v so that $\sigma_{v,i} = \frac{m_i}{\pi R_v^2} 1(\mathbf{r} \in \Omega_i)$ and
 148 $\sigma_v = \sum_i \sigma_{v,i}$ where $\Omega_i = \{\mathbf{r} : |\mathbf{r} - \mathbf{r}_i| < R_v\}$.

149 To solve the Poisson equation, we integrate the Green function $G(\mathbf{r}, \mathbf{s})$ of Poisson
 150 equation with the source.

$$\lambda Z(\mathbf{r}) = \int G(\mathbf{r}, \mathbf{s})(1 + \tilde{P}(\mathbf{s}))d\mathbf{s}^2 \quad (42)$$

$$= \int G(\mathbf{r}, \mathbf{s})d\mathbf{s}^2 + \frac{1}{\sigma} \sum_i \int_{\Omega_i} G(\mathbf{r}, \mathbf{s})\sigma_{v,i}(\mathbf{s})d\mathbf{s}^2 \quad (43)$$

$$\equiv I_1 + I_2 \quad (44)$$

151 where the Green function on a disc with radius R is

$$G(\mathbf{r}, \mathbf{s}) = \frac{1}{2\pi} \log |\mathbf{r} - \mathbf{s}| - \frac{1}{2\pi} \log \left(\frac{|\mathbf{s}|}{R} \cdot \left| \mathbf{r} - R^2 \frac{\mathbf{s}}{|\mathbf{s}|^2} \right| \right) \quad (45)$$

$$G(\mathbf{r}, \mathbf{0}) = \frac{1}{2\pi} \log |\mathbf{r}| - \frac{1}{2\pi} \log R \quad (46)$$

152 Let us consider a field point that is not covered by the vehicles $\mathbf{r} \notin \cup_i \Omega_i$. I_1 is the
 153 solution to the case with uniform load that $I_1 = \frac{1}{4}(|\mathbf{r}|^2 - R^2)$. For I_2 , the source is

154 effectively a point source since the field point is outside the source, so

$$I_2 = \frac{1}{\sigma} \sum_i \int_{\Omega_i} G(\mathbf{r}, \mathbf{s}) \frac{m_i}{\pi R_v^2} \delta(\mathbf{s} - \mathbf{r}_i) d\mathbf{s}^2 \quad (47)$$

$$= \frac{1}{\sigma} \sum_i m_i G(\mathbf{r}, \mathbf{r}_i) \quad (48)$$

155 Up till so far, we have solved the shape of the membrane $Z(\mathbf{r})$. Next, we evaluate the
 156 height of the i th vehicle. Since the vehicle is not a point object, we average the membrane
 157 height Z on the rim of the vehicle to approximate the height of the vehicle z_i .

$$z_i = \langle Z \rangle_{\partial\Omega_i} \quad (49)$$

$$\lambda z_i = \langle I_1 + I_2 \rangle = \langle I_1 \rangle + \langle I_2 \rangle \quad (50)$$

158 $\langle I_1 \rangle$ is contributed by the self weight of the entire membrane so that we approximate
 159 it by just the value at the center of the vehicle \mathbf{r}_i : $\langle I_1 \rangle = \frac{1}{4}(|\mathbf{r}_i|^2 - R^2)$.

160 For $\langle I_2 \rangle$, there are two different types of contributions. The first ones are the patches
 161 of domain from the vehicles other than the i th vehicle, the one of concern that contribute
 162 as far field. The second type is the contribution from the load of vehicle i itself.

163 For the first type, we still use the point source approximation:

$$\langle I_{2,j \neq i} \rangle = \frac{m_j}{\sigma} G(\mathbf{r}_i, \mathbf{r}_j) \quad (51)$$

164 For the second type:

$$\langle I_{2,i} \rangle = \frac{m_i}{\sigma} \langle G(\mathbf{r}, \mathbf{r}_i) \rangle_{\mathbf{r} \in \Omega_i} \quad (52)$$

$$\begin{aligned} &= \frac{m_i}{2\pi\sigma} \left(\langle \log |\mathbf{r} - \mathbf{r}_i| \rangle - \left\langle \log \left(\frac{|\mathbf{r}_i|}{R} \cdot \left| \mathbf{r} - R^2 \frac{\mathbf{r}_i}{|\mathbf{r}_i|^2} \right| \right) \right\rangle \right) \\ &= \frac{m_i}{2\pi\sigma} \left(\log R_v - \log \left(\frac{|\mathbf{r}_i|}{R} \cdot \left| \mathbf{r}_i - R^2 \frac{\mathbf{r}_i}{|\mathbf{r}_i|^2} \right| \right) \right) \\ &= \frac{m_i}{2\pi\sigma} \log \left(\frac{R_v R}{R^2 - |\mathbf{r}_i|^2} \right) \end{aligned} \quad (53)$$

165

Piecing all these terms together, we arrive at the z position of the i th vehicle is

$$\begin{aligned}
 2\pi\lambda z_i &= \frac{\pi}{2}(|\mathbf{r}_i|^2 - R^2) + \frac{m_i}{\sigma} \log\left(\frac{R_v R}{R^2 - |\mathbf{r}_i|^2}\right) \\
 &+ \frac{1}{\sigma} \sum_{j \neq i} m_j \left(\log \frac{|\mathbf{r}_i - \mathbf{r}_j|}{|\mathbf{r}_i - \mathbf{r}'_j|} - \log \frac{|\mathbf{r}_j|}{R} \right)
 \end{aligned} \tag{54}$$

166

where $\mathbf{r}' = (R/|\mathbf{r}|)^2 \mathbf{r}$ is conventionally regarded as the position of the image charge.

167

\mathbf{r}_j 's are the positions of the other vehicles.

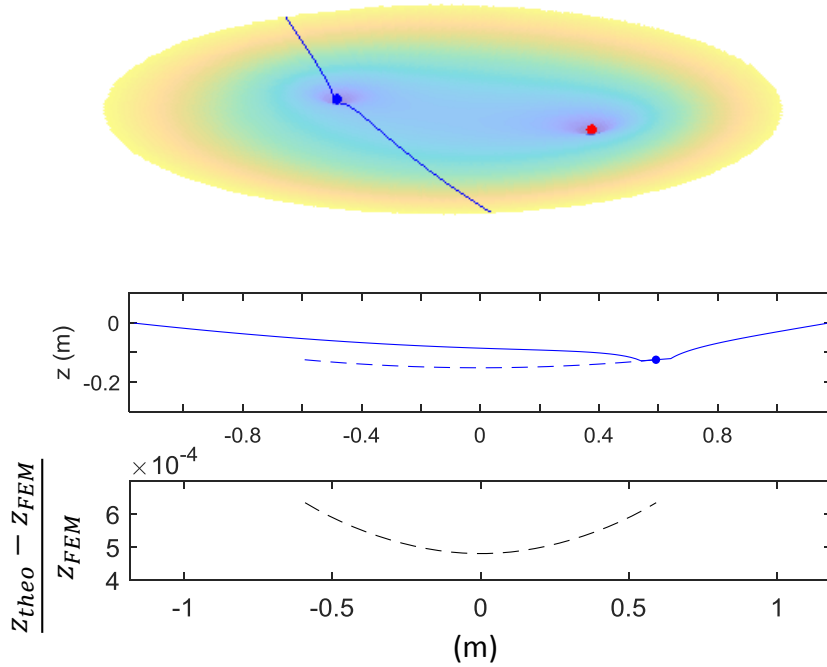


Figure S8: Numerical verification of the analytical solution: We show a test with the blue vehicle put at different y positions while the x position is fixed (0.2 m). The solid blue line shows the membrane shape and the dotted line shows the vertical position of the vehicle z when placed at different positions. The bottom panel shows the relative error of z between the analytical (Eq.54) and numerical (FEM) solution.

168

Despite the fact that some approximations are made, the analytical solution matches

169

with the numerical result (FEM) with a relative error smaller than 10^{-3} (Fig.S8).

170 S8. Dynamics of two vehicles with the same mass

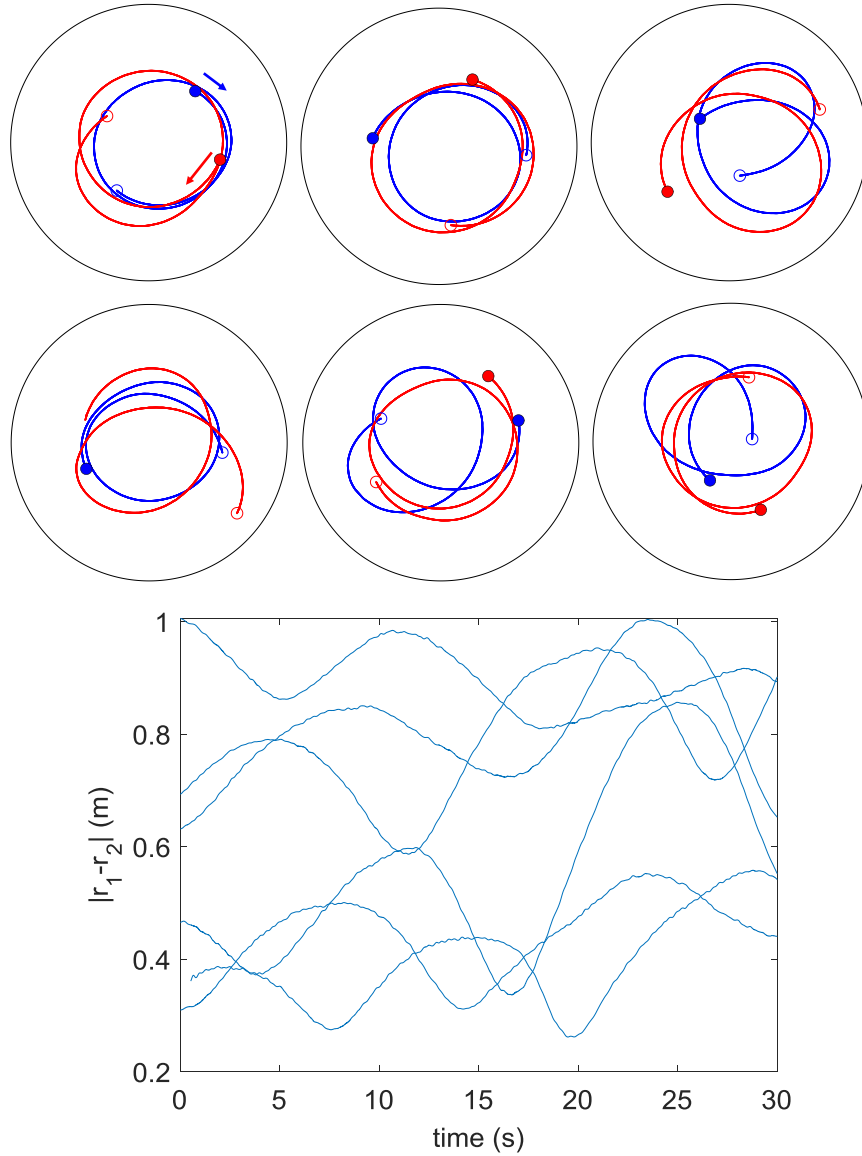


Figure S9: Dynamics of two vehicles with the same mass (a) Trajectories of vehicles with the same mass started at different initial conditions. (b) The relative distance of the two vehicles in (a).

171 **S9. Supplementary movies**

172 **Movie S1: Trajectory of the heavy car (200 gr.) moving on elastic**
173 **membrane: Circular orbit**

174 **A typical circular orbit:** A video of a robotic vehicle driving on an elastic membrane
175 with a central depression of 9.6 cm. Instantaneous velocity and radius (r) are marked with
176 red and green arrows, respectively. The heading angle is the angle between the velocity
177 and radius.

178 **Movie S2: Trajectory of the heavy car (200 gr.) moving on elastic**
179 **membrane: Retrograde precessing orbit**

180 **A typical precessing orbit (retrograde):** A video of a robotic vehicle driving on an
181 elastic membrane with a central depression of 9.6 cm. Instantaneous velocity and radius
182 (r) are marked with red and green arrows, respectively. The tracking shows that the apsis
183 of the orbit is rotating in the opposite direction of the orbit.

184 **Movie S3: Trajectory of the light car (45 gr.) moving on elastic**
185 **membrane: Prograde precessing orbit**

186 **A typical precession orbit (prograde):** The lighter vehicle's orbit undergoes a pro-
187 grade precession, i.e. the vehicle and the periapsis rotate clockwise. The mass of the
188 vehicle is about one quarter the mass of the vehicle used in Movie S1 and S2. As pre-
189 dicted by the theory, a radial attraction $k(r)$ decreasing with r enabled by a lighter vehicle
190 leads to the precession with the same sign of orbit, as opposed to the precession in Movie
191 S2.

192 **Movie S4: Deformation-only induced motion**

193 In this movie, the membrane deformation is created by a human-controlled meter stick.
194 The motion of the vehicle re-oriented by this deformation shows the deformation itself
195 can act as a force to affect the motion of an object on the membrane.

196 **Movie S5: Deformation-induced merger**

197 In the first part, both panels show the trajectories of two vehicles moving on the mem-
198 brane at the same time. The comparison is made regarding the mass ratio between the
199 two vehicles: when the leading vehicle is heavy enough ($m_{21} = 1.37$), the two vehicles
200 eventually merge while the $m_{21} = 1.00$ fails to merge. In the second part, the video on
201 the right panel shows the virtual superimposition of independent runs of the two vehicles
202 with the same mass ratio as the left panel shows that the substrate-mediated interaction
203 is indeed making the two vehicles interact.

204 **Movie S6: Controlling speed with tilt angle to avoid collisions**

205 Each video shows the trajectories of the IMU-controlled vehicle (white chassis, solid line)
206 and uncontrolled vehicle (gray chassis, dashed line) when a particular control magnitude
207 $A = 0, 2, 4, 8$ is used.

208 **Movie S7: Controlling speed with tilt angle to avoid collisions**
209 **(Simulations of 5 vehicles)**

210 The movie shows the simulations of 5 vehicles moving on the same membrane as in
211 experiments. Starting from the same initial condition, the vehicles applied with speed
212 control scheme avoid from collisions while the ones without merge quickly. More details
213 can be found in Fig.9 of the main text and the corresponding section.